

Uncertainty in Damage Assessment and Remaining Life Prediction of Engineering Materials Used In Petrochemical Industry

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Abstract

In this paper creep damage assessment of about 11 years' service exposed HP-40 grade of steel used in hydrogen reformer of a petrochemical industry has been carried out in terms of a discontinuous Markov process. Experimentally determined conventional creep data under identical testing condition were used in the present investigation. Scatter and damage accumulation due to creep deformation were evaluated through microstructural assessment using light optical microscope and scanning electron microscope. Quantification of creep damage was made from replicated creep data in terms of two damage parameters A and A*. Statistical analysis of void area fraction has been carried out extensively for the both top and bottom portions of the reformer tube at 870°C in the stress range of 52-68 MPa. In addition, the proposed probabilistic model has been compared with the Kachanav's Continuum Damage Mechanics (CDM) model. Both the approaches displayed quantitative experimental support. A residual life of > 10 years is estimated at 870°C/ operating stress. For 55 years' service exposed Catalytic Cold Cracking (CCU) reactor vessel and Feed Processing Unit (FPU) distillation column materials of a petrochemical industry remnant life assessment studies were estimated by incorporating the uncertainty involved in calculation of LMP (Larson Miller Parameter) values and from extrapolation of stress vs. LMP plot. Variability of normalized creep damage for reactor and column materials is well approximated with the aid of Weibull distribution. As expected, it is observed that the distributions shift towards the higher range of damage with increase in service exposure time.

Keywords: Conventional creep test; Void fraction; Damage assessment; Fractography; Microstructural assessment; Statistical analysis; CDM model; Larson–Miller parameter

Introduction

HK and HP series are centrifugally cast austenitic stainless steels, widely used as for reformer tubes in the petrochemical industry for ammonia, methanol and hydrogen plants. And they operate above 900°C. Amongst them HP40Nb and micro alloyed is more popularly used around the world for such application. The main criteria for their selection are their high creep strength at and above 900°C [1-3]. But in remaining life estimation studies of these materials, scatter characteristic of service exposed material is not taken into account for damage.

Under service exposure condition, reformer tubes have been subjected to carbonization, oxidation, overheating, stress corrosion cracking, sulfidation and thermal cycling. Overheating of the tubes leads to early creep damage and embrittlement [3-7]. Premature failure of these tubes can take place due to service ageing which is the major cause for rupture Traditional stainless steel application for vertical reformer tubes has been gradually replaced or substituted by centrifugally cast HK40 alloys and subsequently by HP modified alloys. Even after these advancements, many reformers fail during service within a period of 3 to 15 yrs, though their designed performance is specified as 11.4 yrs. On the other hand, it is noteworthy that despite most of these tubes have a specific design life of 10-15 yrs, literature survey has revealed that they can have a significant remnant life beyond their specified design life [2]. This is estimated by carrying out RLA (residual life assessment) studies performed at progressive intervals during a well-planned shut down, both by non-destructive evaluation technique and also by destructive tests like accelerated creep or stress rupture tests above the operating

hoop stress level of the tubes. Balance or residual life of the tubes can then be estimated by using LMP (Larson-Miller Parameter) equation [5].

Some past observation of remnant life on service exposed materials has been reported in literature [2,4-7]. But in these studies scatter characteristic of service exposed material is not taken into account for damage/ remaining life assessment. The Weibull distribution function [8,9] is commonly used to model the statistical behaviour of experimental results to provide a reasonable fit to the probability distributions of life times (time to rupture) obtained from such tests. This explains to some extent the variability in life time. However the probability distribution of damage as it evolves during the test cannot be reasonably represented by Weibull distribution in many cases. This is because it looks only at two limiting states, viz., and the initial state when material is virgin and the final state when the material has failed (ruptured). This imposes severe limitations as it cannot predict any

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intermediate states of damage accumulation. Bogdanoff and Krieger [10] and Ganesan [11] have developed an evolutionary probabilistic model for fatigue life time prediction. This gives the evolution of the probability distribution as a function of number of cycles or time.

Most structural materials are developed by optimizing microstructure to deliver a specified short term mechanical performance. During high temperature service, this carefully tailored microstructure may progressively degrade with often unexpected consequences for long term creep performance. An effective life assessment strategy not only requires a robust creep mechanics framework but also a quantitative description of the various microstructural damage mechanisms. Therefore it is essential to quantify the microstructural damage parameters that have a significant influence on creep behaviour. Nucleation and growth of cavities or voids at grain boundaries or even at precipitate/matrix interface are responsible for creep rupture of the tubes. Quantification of creep damage was made from replicated creep data in terms of two damage parameters A and A*. Nucleation of cavities at grain boundaries & precipitate matrix interface and their subsequent growth & coalescence are thought to be the mechanisms for its creep rupture. A suitable methodology of characterizing creep damage evolution and quantitative metallography evaluation of creep cavitation has not so far been studied.

In the present work, Bogdanoff model (B-model) based on Markov process [12] and Kachanov [13] CDM model (K-model) have been used to predict creep damage evolution. Monte Carlo Simulation has been carried out to predict the associated scatter in damage evolution under identical test conditions. The model(s) have been applied to creep strain accumulation data of 11 years' service exposed reformer tube (both top and bottom portion) made up of HP40 grade of steel in a petrochemical industry, for different stress-temperature conditions. Also a quantitative metallography study on the extent of creep cavitation and statistical analysis of void area fraction has been carried out extensively for both top and bottom portion of the reformer tube at 870°C in the stress range of 50-68 MPa. The method of quantitative metallography evaluation of voids on crept samples of this material has already been reported elsewhere [14].

Material and Experimental Procedure

The tube material used in this study is made of HP40 grade of steel which is about 11 years' service exposed and operating at 870°C, though the designed operating temperature of the tube specified by the plant is 920°C. The tubing material is equivalent to 25Cr35Ni1Nb0.2Ti0.01Mo. Room and high temperature (850°C, 870°C and 890°C) tensile tests of the material were carried out as per ASTM-8M standard, which enables one to fix the operating stress level during creep tests so that rupture occurs reasonably within a stipulated time frame. Conventional creep tests of the tubes at 870°C were conducted as per ASTM E

139/83 specification in single point ATS Creep Testing Machines (2T capacity), equipped with a three zone split furnace and maintaining zonal temperature within ± 2°C with a high precision controller. Strain measurements were made as per ASTM E 139/83 standard procedure, with LVDT (Linear Variable Displacement Transformer) attachment in the bottom pull rod, outside the furnace but connected to both the extensometers fixed at the top as well as bottom portion of the specimen gauge length, well within the furnace of the ATS creep testing machine. Round bar cylindrical solid specimens having a gauge diameter of 5.0 mm and a gauge length of 28.47 mm were chosen both for tensile as well as creep tests [14]. Creep specimens were loaded at 50-68 MPa / 870°C. The operating hoop stress σ_h (13.46 MPa), was evaluated using the following formula:

$$\sigma_h = PD / 2t \quad (1)$$

Where P is the operating pressure in MPa, D is the mean diameter in mm and t is the thickness of the tube in mm. The stress rupture data have been plotted in terms of stress versus LMP (Larson–Miller parameter):

$$LMP = T (C + \log t_r) \quad (2)$$

Where T is the absolute temperature in K, t_r is the rupture time in hours and C is a constant generally taken as 20. For this grade of steel two values of C (C=15 and C=20) were considered (Table 1) for balance or remaining life prediction as the major alloying element in this grade of steel is Ni.

From Equation (2) rupture time can be specified as

$$t_r = 10^{\frac{LMP}{T} - C} \quad (3)$$

In case of CCU reactor and FPU distillation column materials, accelerated creep rupture tests were carried out in the temperature range of 515-520°C and in the stress range of 60-120 MPa. The stress levels above the operating stress at each temperature were selected in such a way as to obtain rupture within a reasonable span of time. In general, creep rupture damage (ω) is considered as the ratio of cumulative operating time and total operating time [15].

$$\omega = \frac{t}{t + t_r} \quad (4)$$

Creep rupture damage (ω) has been evaluated in terms of Larson–Miller parameter with a functional relation is derived as follows [15]:

$$\omega = a_0 + a_1 LMP + a_2 LMP^2 \quad (5)$$

This relation is true only when one could generate a large number of experimental accelerated creep rupture data with specimens removed during large number of plant shut downs i.e. with various years of

Component (service exposed)	Hoop Stress, MPa	Temp., °C	C (LMP constant)	LMP of Service exposed material	Life (t), yrs	Remaining Life (RL)*, yrs
Reformer tube (Top)	13.46	870	20	30800	> 10	> 10
			15	24100	> 10	> 10
		920	20	30800	> 10	> 10
			15	24100	18	7
Reformer tube (Bottom)		870	20	30600	> 10	> 10
			15	24100	> 10	> 10
		920	20	30600	> 10	> 10
			15	24050	17	6

* RL = $t_r - 11$ (service exposure in yrs)

Table 1: Remaining life at different temperature with varying C (LMP constant) value.

service exposure [15] as in the case of 55 years' service expose CCU reactor and FPU distillation column materials. The column is made of mild steel conforming to ASTM a 285 Gr. C and the reactor vessel conforming to ASTM a 201 Gr. A.

To reduce error in damage calculation, experimental creep damage was normalized by predicted creep damage (Equation (5)). Variability of normalized creep damage can be well approximated with the aid of Weibull distribution [16]. Creep damage (ω) in terms of probability of failure and LMP can be expressed as:

$$\omega = a(a_0 + a_1LMP + a_2LMP^2)((-\ln(1 - P_f))^\frac{1}{b})$$

Where a, b are Weibull parameter and P_f is probability of failure.

In the present study, a set of four specimens were tested under identical condition to examine the statistical creep behaviour. Scatter in voids was quantified from Light Optical Microscopy (LOM), SEM (Scanning Electron Microscope) equipped with image analyzing technique and EDX (Energy Dispersive Xray Spectroscopy) analyser. Identification of various elements was carried out with EDX. A detailed analysis of this technique and its interpretation is given elsewhere [14]. Fractographic features were recorded and studied for all the crept samples in the SEM. One half of the creep ruptured pieces were cut longitudinally along the mid plane, mirror polished

through conventional metallographic technique to reveal creep cavity. Fractography was carried out on the other half of the fractured specimens. Quantification of damage in terms of voids was made both at perpendicular as well as parallel directions to the loading axis of the gage length of the specimens [14]. These observations were made on a plane which corresponds to the mid-section of the specimen diameter at different distances from the ruptured surface. Ten optical images at diametric locations were recorded digitally at 400X. Only after appropriate grey-thresholding, these optical images had been analyzed with a view to obtaining the void area as a function of true strain after creep rupture.

It is well established that true stain is a function of the diameter (D) of the specimen at the specific transverse plane and is represented by [17]:

$$\varepsilon = 2Ln\left(\frac{D_0}{D}\right) \tag{7}$$

Where D is the measured diameter and D_0 is the original diameter of the specimen. For the purpose of maintaining uniformity, void area was normalized with respect to true strain which has been measured from crept and ruptured specimens as shown in Figure 1.

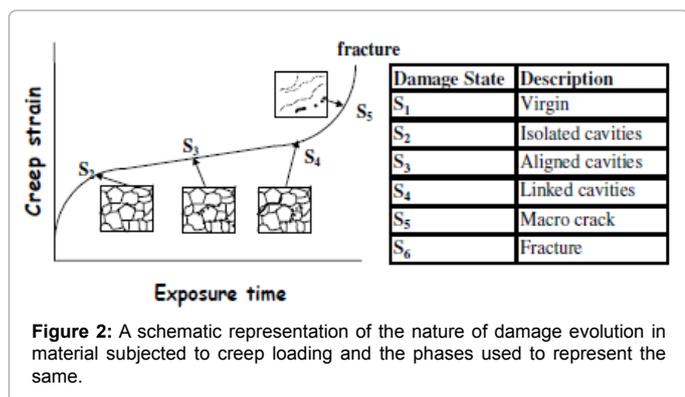
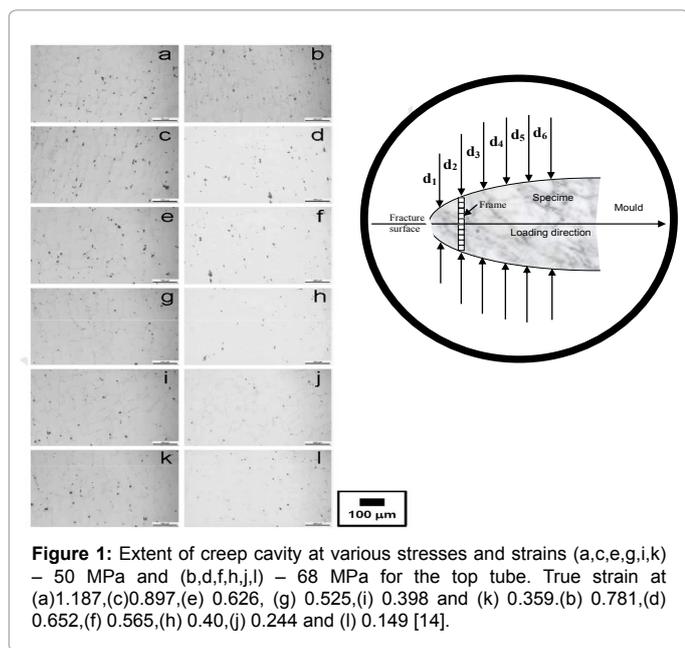
Creep Damage Modelling for Service Exposed Reformer Tubes

In polycrystalline metals undergoing creep at high temperatures, failure occurs mainly by the nucleation and growth of microscopic cavities on the grain boundaries. Cavitation is one of the most common forms of creep damage. This has been correlated with creep stain accumulation by direct metallographic examination of power plant components exposed to creep. Figure 2 shows a schematic representation of the nature and distribution of cavities during various stages of creep strain accumulation in a component. The process truly consists of both nucleation and growth, the former is probabilistic whereas the latter is deterministic. A probabilistic mechanism is best represented as a discrete process whereas growth is continuous by nature. Predictive models based on deterministic approach consider constrained growth of cavities as the principal mechanism. Such models assume presence of extremely fine cavities as pre-existing nuclei. Their initial sizes are too small to account for any loss of effective load bearing cross section area of the component. If at all it contributes it is likely to affect the initial strength of the material and its effect would show up as scatter in the experimental creep strain time plot. Nevertheless with this assumption deterministic model does describe many of the features of the creep curves described above. Following the same analogy it is possible to build up a model entirely based on probabilistic approach assuming creep damage accumulation as a discrete process.

Deterministic approach: CDM model for creep damage of service exposed reformer tubes

The work of Kachanov [13] and Rabotnov [18] is of considerable relevance to evaluate material degradation due to creep. They assumed that during creep, damage accumulation progressively reduces the effective load bearing cross sectional area of a specimen from an initially intact condition to a complete loss when rupture occurs. Kachanov [13] CDM model (K-model) has been used to estimate creep damage. The model is well established and has been applied to analysis of accumulation of creep strain data for 11 years' service exposed reformer tube (both top and bottom portions) for different stresses at 870°C in the present investigation.

Strain rate accumulation can be predicted throughout the creep



life by introducing a generalized damage parameter (ω). Estimation of damage or loss of continuity is made possible by a set of two coupled equations which fit the limiting conditions i.e., $\omega = 0$ at time $t = 0$ and $\omega = 1$ at time $t = t_r$ as:

$$\frac{d\varepsilon}{dt} = \frac{A\sigma^n}{(1-\omega)^n} \quad (8)$$

$$\frac{d\omega}{dt} = \frac{B\sigma}{(1-\omega)} \quad (9)$$

Where, A, B, n, m and η are material constants.

Following Cane [19], Equations (8,9) can be integrated for constant stress conditions to give

$$\omega = 1 - \left(1 - \frac{t}{t_r}\right)^{\frac{1}{\lambda}} \quad (10)$$

Where, λ is creep damage tolerance factor [19-22] and is defined as the ratio of strain to failure to the Monkman-Grant constant i.e., $\lambda = \varepsilon_r / \varepsilon_m t_r$ and $\varepsilon_r, t_r, \varepsilon_m$ are rupture strain, time to rupture and minimum creep rate respectively. The product $\varepsilon_m t_r \equiv \varepsilon_s$ is the Monkman-Grant constant. λ is a significant parameter in assessing the susceptibility of a material to localized cracking at strain concentrations and is also suggested as a better measure of creep ductility [20-22]. Its value for engineering alloys ranges from 1 to about 20.

Nucleation of cavities being deterministic, the scatter band in damage evaluation under identical test conditions can be easily estimated by carrying out a Monte Carlo Simulation. Similar simulation study was carried out on service exposed austenitic stainless steel [23] and P22 grade of steel [24].

Equation (10) defines a generalized creep damage relationship for estimating creep life fraction consumed at any given time. However,

this approach does not consider the scatter that is inherent in any creep test. It would be highly constructive if estimation of remaining life of component could effectively be positioned within the scatter band of creep strain vs time plots.

Stochastic approach: Markov process for creep damage of service exposed reformer tubes

Nucleation of cavities is one of the principal mechanisms of damage. It considers growth as nucleation on the interface between a pre-existing cavity and the matrix. Growth of cavities/voids is probabilistic. With creep damage accumulation the distance between cavities keep decreasing and ultimately some of these may join together to produce a micro crack. This happens as and when a cavity nucleates in between two adjacent cavities. The process continues leading to the formation of macro cracks. Ultimately the component fails as and when its load bearing cross section area is full of cavities/cracks. In fact Figure 2 gives a schematic representation of the above stages of nucleation and joining of cavities leading to fracture at different levels of strain accumulation. A virgin material passes through each of these stages before failure occurs. The accumulation of micro cracks inside the material is considered as a cumulative damage process. The process of damage accumulation during creep depends only on its current state. Therefore it is possible to simulate this by a Markov chain process.

The Markov process is defined as a stochastic process satisfying the following condition:

$$\text{Prob}[S_{n+1} = j | S_n = i, S_{n-1} = k, \dots, S_0 = l] = \text{Prob}[S_{n+1} = j | S_n = i] \quad (11)$$

The indices i, j, k , etc. stand for the state number and S_n stands for n^{th} event. Equation (11) says that probability of S_{n+1} being in state j is only affected by the value of S_n and the history of state value has no effect on S_{n+1} .

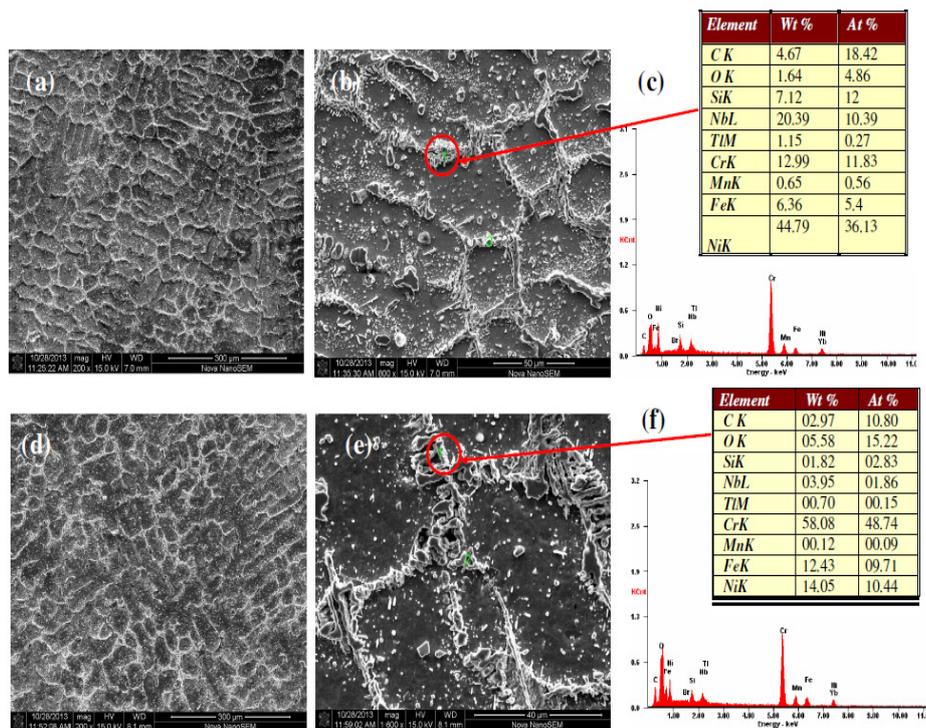


Figure 3: SEM micrograph revealing microstructure of as received top and bottom portions of the reformer tube with EDAX analysis of Cr and Nb rich precipitates.

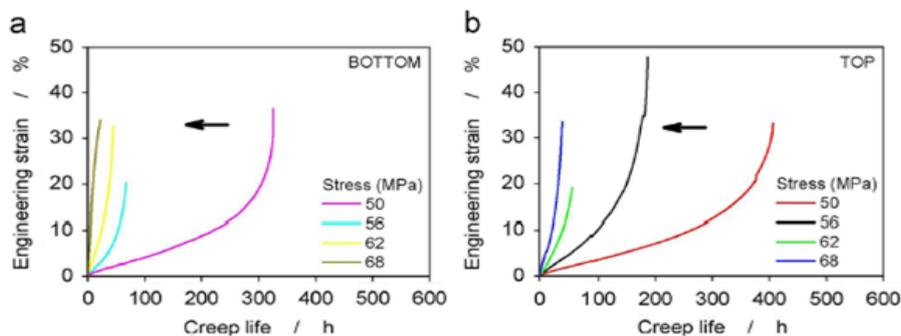


Figure 4(a): Experimental creep curves at four different stress levels for the (a) bottom and (b) top tubes.

State **1** indicates the initial state where there is with no damage and state **b** indicating the final damage state (rupture or failure state) or the absorbing state. Leckie et al. [22] have defined the initial distribution of the different damage levels at time $t = 0$, by a $(1 \times b)$ row vector p_o :

$$p_o = \{ \pi_1, \dots, \pi_b \} \text{ and } \sum_{j=1}^b \pi_j = 1 \quad (12)$$

Probability of moving to state 3 and beyond is zero [23,24]. Accumulation of creep damage in a material is represented by a transition matrix P , [25]. Please refer to Appendix I for details.

All states in the transition matrix P are transient, except for state **b** (absorbing state). Bogdanoff and Krieger [10] had suggested a simple procedure for estimating these elements to simulate fatigue damage evolution. The same procedure had been used in this work as $B -$ model to predict creep damage evolution as a function of time.

Statistical analysis of creep cavity area for service exposed reformer tubes

Most of the efforts in the study of creep in steels have focused on experimental research to examine the effect of microstructure on the formation of cavities. Nucleation of cavity is a random process and is strongly dependent upon the local stress-state and strain-state. It has been observed that over various test conditions, for the both top and bottom portion of the reformer tube the normalised void/ creep cavity area fraction, $F(V_f) = \left(\frac{v_f}{\epsilon} \right)$ follows the Weibull distribution. To maintain uniformity, creep cavity area was normalized by respective true strain which has been measured from failed crept specimens as shown in Figure 1. In terms of true strain (ϵ) and cumulative probability of failure P_f , void area fraction (v_f) can be probabilistically estimated as [8]:

$$v_f = a\epsilon \left(-\ln(1 - P_f) \right)^{\frac{1}{b}} \quad (18)$$

Where a and b are Weibull parameter.

Results and Discussion

Microstructure evaluation

Metallographic examination under SEM is revealed in Figure 3. The as received (uncrept) service-exposed reformer tubes showed that the material did not possess any sign of degradation including decarburization and voids or creep cavitation. Microstructural examination under SEM is revealed in Figure 3a-3d). Both the top as well as bottom portions of the reformer tube revealed columnar type of microstructure (Figure 3a,3d). In general, a columnar type of structure was observed for both the top and bottom portions of the reformer tube. Basically two types of carbides are revealed in the EDX analysis- (a) the primary carbide network consisted of chromium-rich $Cr_{23}C_6$

and (b) niobium-rich carbides as intra-dendritic carbides (light grey ones), with presence of some nickel and silicon in it (Figure 3). Nb rich carbides appear to be coarsened and coagulated. Thus, one could see various morphologies of the precipitates in the intra - dendritic region.

Formation of coarse carbides at dendritic boundaries in the bottom region of the tube as well as gradual disappearance of fine intra-dendritic precipitates were mainly due to overheating (Figure 3). Due to ageing, these carbides would coarsen and shape themselves as worm [2,26]. Presence of significant amount of silicon and nickel at the light grey precipitate confirms that partial conversion of NbC phase to Nb-Ni-Si had occurred [2]. The main damage mechanism for reformer tube is the combination of thermal stresses and internal pressure stresses [2,26]. This combination of stresses causes creep damage that typically develops at the inner diameter or just below the internal diameter surface. It was pointed out that test stress was different from service stress because various pressure and wall thickness combination did exist.

Conventional and accelerated creep rupture behaviour of service exposed reformer tubes

From the hot tensile strength data [14] creep loads at 870°C were selected in the range of 50-68 MPa for the service exposed reformer tubes. Hardness and tensile test results [14] when compared with the standard NRIM (National Research Institute for Metals) data of similar grade of steel [27] has clearly established that the material has become softened due to prolonged service exposure, which is also apparent from presence of coarse carbides at dendritic boundaries and disappearance of fine intra dendritic precipitation (Figure 3), at the bottom portion of the tube. Figure 4i represents the experimental creep curves at four different stress levels for the (a) bottom and (b) top tubes. All these curves clearly depict absence of the primary or Stage I creep region.

The conventional creep curves at 870°C (Figure 4ii) and creep strain rate curve (Figure 4iii) of both top as well as bottom portion of the reformer tube corresponding to various stress levels (50 MPa, 56 MPa, 62 MPa and 68 MPa) clearly exhibit that there is substantial amount of scatter in creep deformation and creep strain rate curve. It should be noted that no valid service history records were available for the individual tubes; thus, it is possible that the current test samples were machined from tubing which was replaced during an earlier shutdown of the refinery i.e. 5 years ago.

Figure 5 exhibits the fractograph of crept service exposed specimens. The fractured surface (Figure 5c,e,g,k,p) has undergone severe oxidation. It was found to rupture in a relatively brittle manner with 16.4% elongation for top portion of the tube. Figure 5a-5p is a

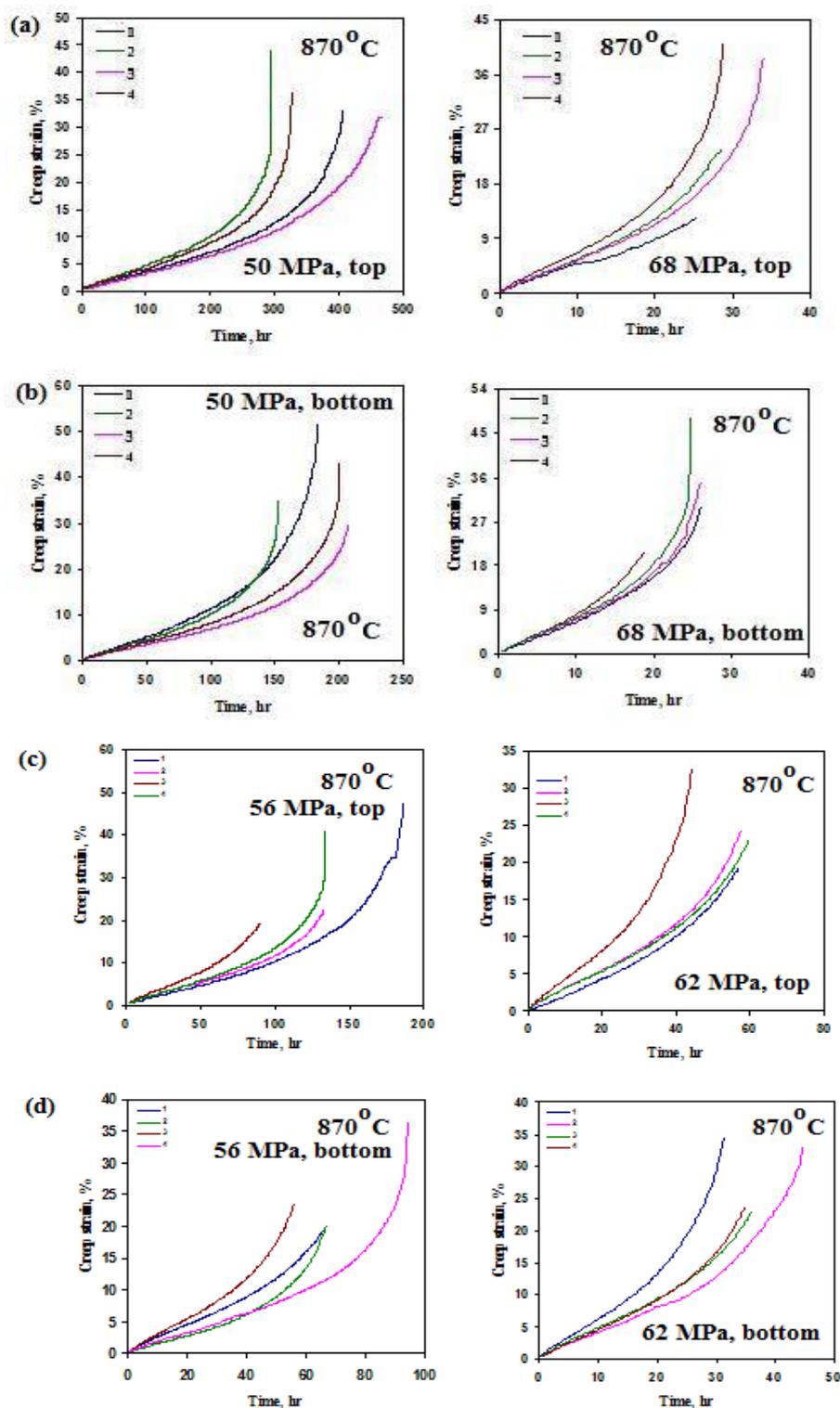


Figure 4(b): Scatter creep curves at 50 MPa and 68 MPa respectively: (a) top and (b) bottom portion of service exposed reformer tube.

general observation of fractographic features of the top as well as the bottom portion of the tube. Bottom portion of the tube fractured with 17% elongation revealing a quasi-cleavage fracture (Figure 5c,d,g,k,o,p) and a brittle failure with presence of cleavage facet (Figure 5g,o,p)

respectively. However, in the top section (Figure 5a,b,f,i,j,n) and Figure 5d,h,l of the bottom section did reveal ductile dimples. In case of quasi cleavage fracture (Figure 5c,d,g,k,o,p) possibly, when the specimen finally fractured, the crack propagates into the specimen and remaining

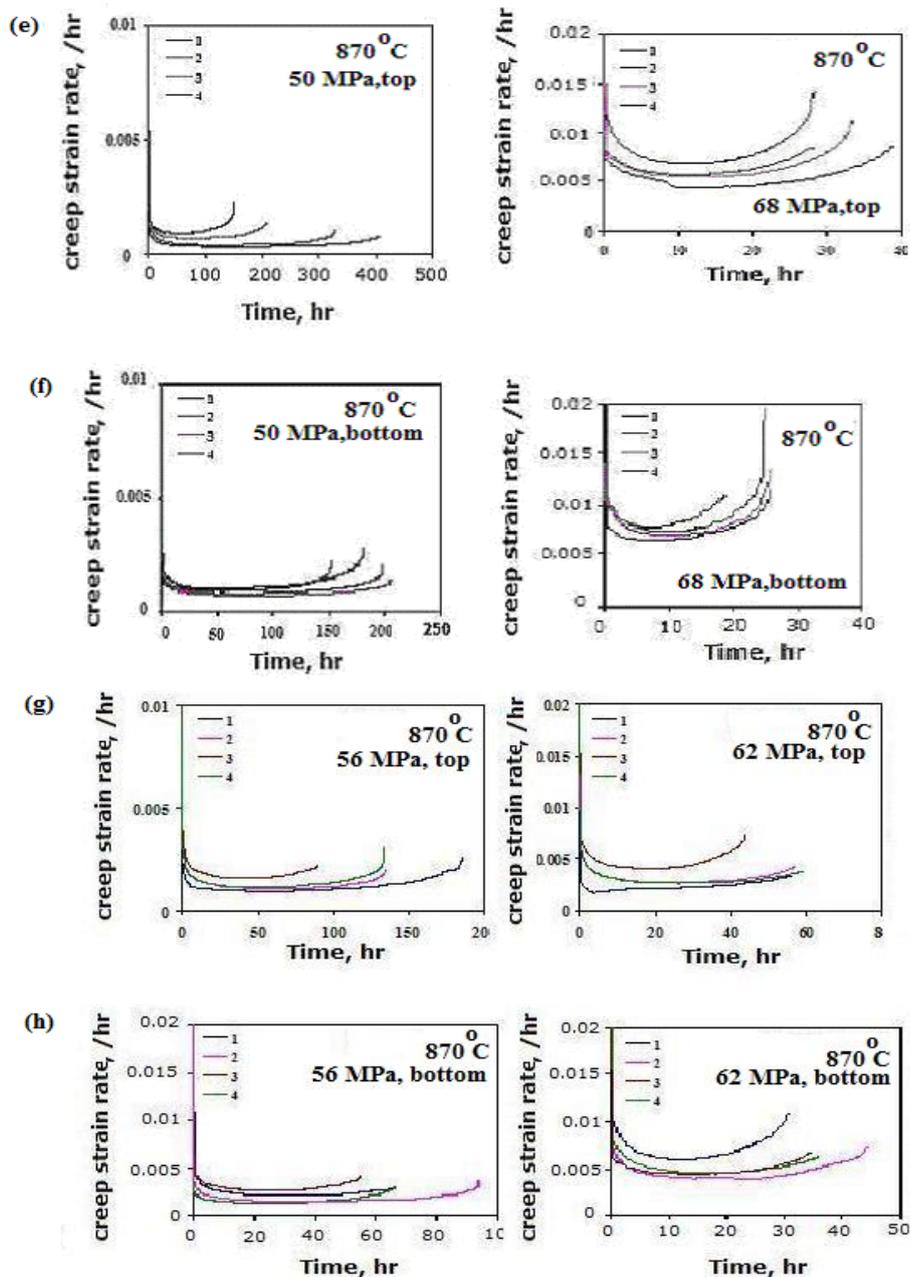


Figure 4(c): Scatter creep strain rate curves at 50 MPa and 68 MPa respectively: (e) top and (f) bottom portion of service exposed reformer tube.

part of the specimen supports the applied load and finally ruptured with shear mode or with sliding. Actually, the facet looks smeared by sliding. Although, samples were crept under identical stress-temperature condition, it is observed that mode of failure is different. Similar kind of observation was found by Garofalo et al. [7] and Roy et al. [28].

For crept service exposed material, fracture was mainly due to the creep rupture mechanism in tertiary region which varied from ductile to quasi cleavage failure. Possibly, as a result of this variation in the mode of fracture, there is a substantial amount of scatter, both in the creep response and creep rupture time. Oxide scale formation (Figure 5e,i,g,k) decreases the effective cross section area for the load

bearing capacity of the specimen. Variation in the mode of fracture may be due to oxide scale formation on the crept specimen which led to catastrophic failure. In addition, though oxide scale reduces the effective cross sectional area of the specimen, the combined effect of inhomogeneity in microstructure of service exposed material is equally important. Hence, one may say that oxidation effect in combination with the influence of change in microstructure could be responsible for variation in the mode of fracture and creep response.

Quantification of creep cavities after rupture

Population of voids (appearing grey) at various stresses and strains is shown in Figure 1. The population of void cavity on the creep ruptured

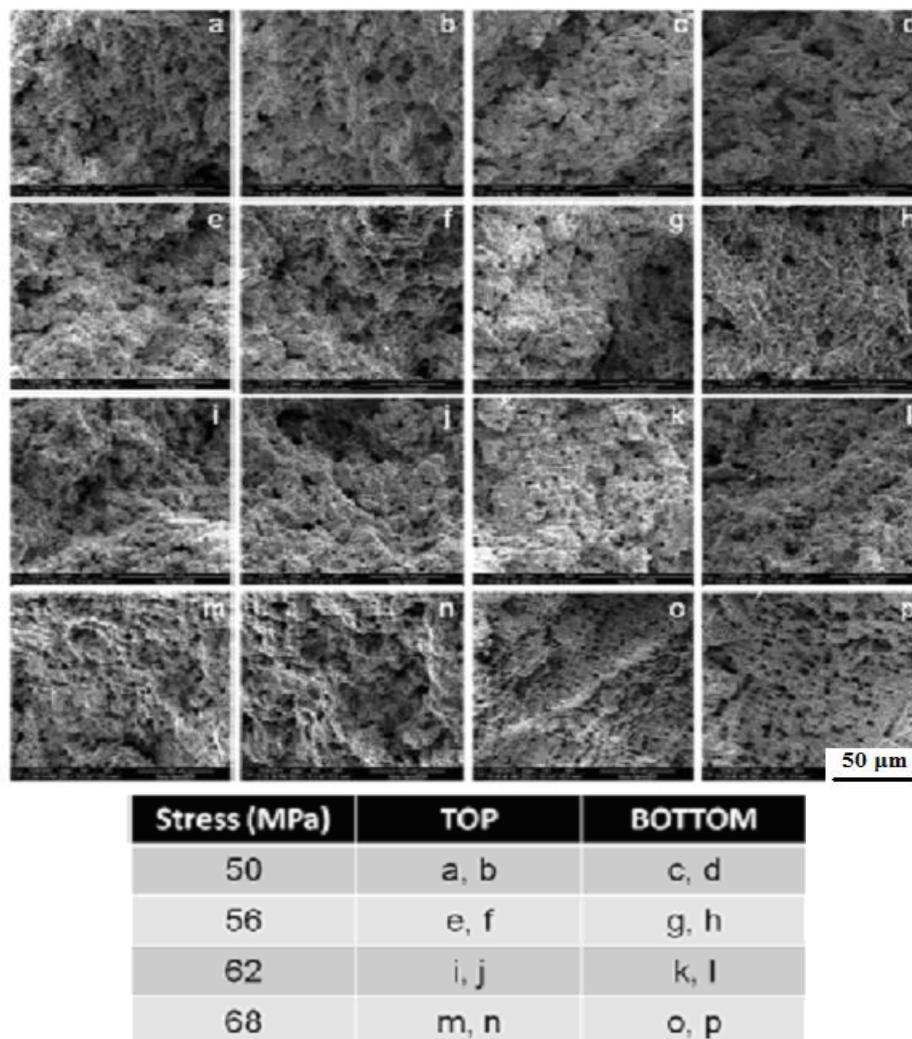


Figure 5: Fractograph of crept service exposed specimens (both top and bottom portion of the tube) tested at 870°C in the stress range of 52-68 MPa.

surface was found to be nucleated either at Grain Boundary (GB), or at Grain Boundary Triple Point (GBTP) predominantly and also in the vicinity of other carbide particles. A similar kind of metallographic technique was employed to measure the extent of marten site at various strain reported elsewhere [17], which is already well established.

There is a systematic correlation of void/creep cavitations and the applied stress at GB (grain boundary) and GBTP (grain boundary triple point). It was found that the probability of GBTP cavitations [14] was almost three times than that of GB voids (Figure 6) where the initial inclusion volume fraction all the specimens remained unaltered. Void distribution at GB and GBTP was inhomogeneous in the deformed austenite dendrites. The damage evolution law for A^* can be obtained by relating A^* to A . The A - parameter which is equated to the damage parameter ω , from K-model has been related to the measured parameter A_1 as following [29]:

$$A = \omega = A_1 \times 4/\pi \quad (19)$$

Where A_1 is number fraction of cavitated grain boundaries on the reference line.

Appearance of voids at GB and GBTP along a reference line is revealed in Figure 6a). Since quantification of voids in the present crept material has shown (Figure 6b,6c) that the probability of number of creep cavities or voids at the GBTP(A_2) junctions along a reference line, is three times than that of the number of creep cavities or voids occurring along the GB (A_1) [14], a relation A_1 and A_2 can be obtained as:

$$A_2 = 3 A_1 = 3 A^* \pi/4 \quad (20)$$

Cavitated GB and GBTP counted for the determination of A_1 and A_2 parameters were measured according to Riedel [30]. GB and GBTP were considered to be damaged when it contained cavity of size greater than or equal to 1 μm . Using the stereological results from Eggler et al. [29], A^* is derived as the cumulative contribution from damaged grain boundaries (A_1) and number of creep cavities or voids at the grain boundary triple points (A_2). Therefore, we arrive at the following:

$$A^* = A_1 + A_2 = \pi A/4 + 3\pi A/4 = \pi A \quad (21)$$

It is more appropriate if 3D measurements of creep cavitation could be carried out. However, due to experimental limitation 2D measurement was resorted to, extensively.

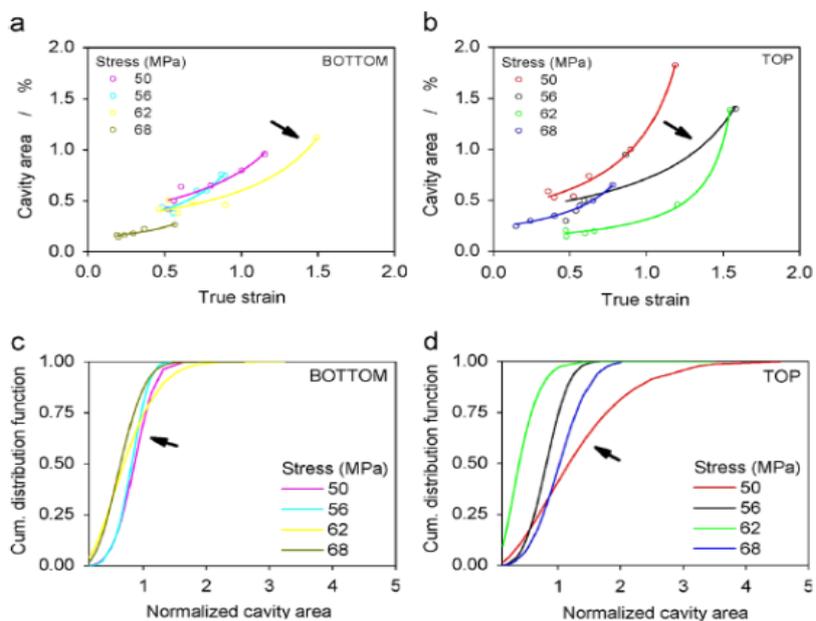


Figure 6: Cavity area as a function of strain at different stress levels: (a) bottom and (b) top tubes. Cumulative distribution functions as a function of normalized cavity area (cavity area/true strain) for (c) bottom and (d) top tubes [14].

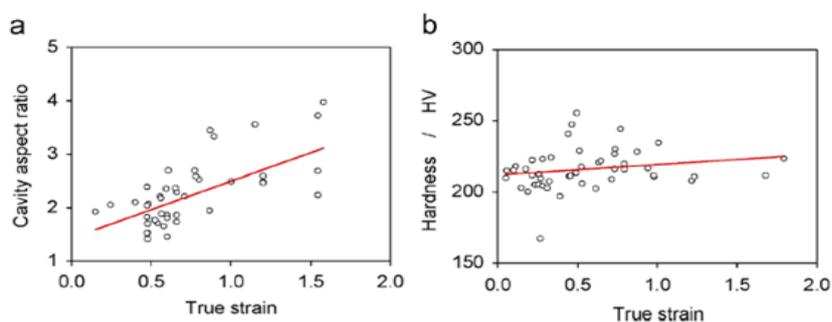
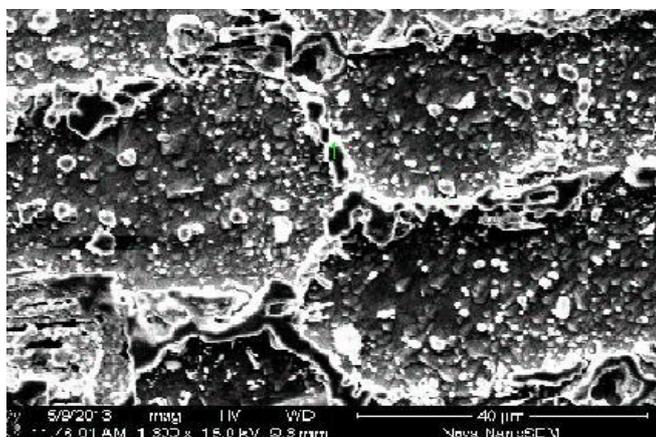


Figure 7: Cavity aspect ratio and (b) hardness as a function of strain for all the stress levels for both top and bottom tubes together [14].



(a)

Figure 8a: (a) SEM micrograph at 50 MPa/870°C, revealing creep cavitation/void formation at GB (white arrow) and at GBTP (black arrow) which were digitally recorded along a reference line indicated in white.

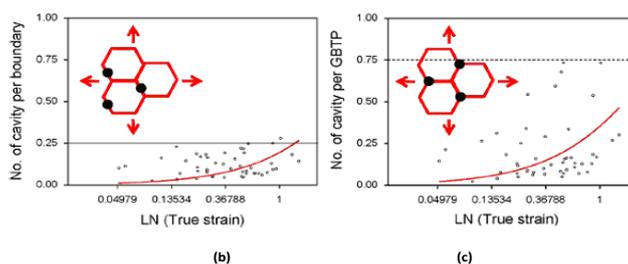


Figure 8b: (b) Number of creep cavity formation at GB and at (c) GBTP as a function of strain for all stress levels for both the top and bottom tubes together [14]. The lines in the graphs indicated the upper boundary limit of the voids/cavity. The best fit curve for number of creep cavity is represented in Figure 8 (b & c).

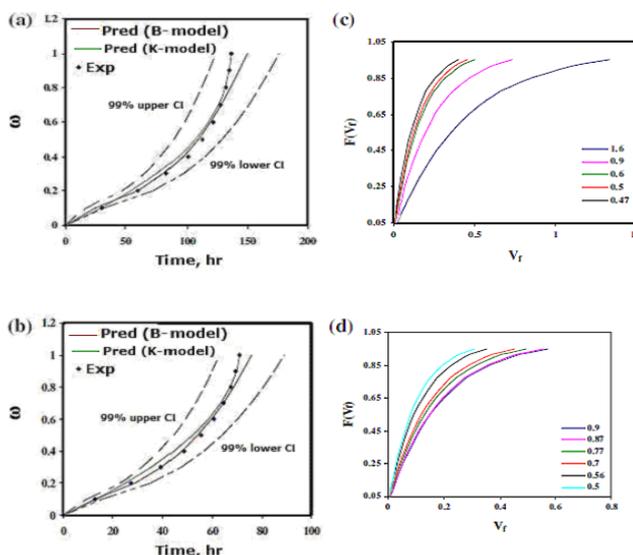


Figure 9a: Comparison of the experimental and predicted values of mean time to reach a specified damage state by both Bogdanoff model and CDM approach with relative Confidence Interval (CI) at 870°C/56 MPa: (a) top and (b) bottom portion of service exposed reformer tube. Probability distribution of void area fraction at 870°C/56 MPa: (c) top and (d) bottom portion of service exposed reformer tube.

Damage modelling for service exposed reformer tubes

In this section, the proposed model based on Bogdanoff model [10] is compared with Kachanov [13] CDM model from both phenomenological and statistical points of view. The phenomenological comparison helps to investigate whether the proposed model could predict the time to reach a specified damage state. Figure 9(a,9b) shows the comparison of the proposed model with the Kachanov's damage model. The proposed model fits the Kachanov's damage model very well, and it can describe the sudden changes of the creep damage in the tertiary part as well.

The Bogdanoff model [10] has been used to predict creep damage evolution as a function of time. Since the probability of reaching a specific state of damage at any instant of time is known, a Monte-Carlo simulation has been carried out to estimate the time to reach a specific damage state. Also, the Kachanov [13] and Rabotnov [18] model has been used to predict creep damage evolution. A comparison of the experimental and predicted values of mean time to reach a specified damage state by both Bogdanoff model and CDM approach with relative Confidence Interval (CI) is shown in Figure 9(a,9b) for both top and bottom portion of the reformer tube. The two extreme lines contain

the range in which 99% of all mean times to reach a specified damage state will lie with 99% certainty. It is of prime importance to mention that the damage prediction by both the model is in close agreement with experimental data.

The probability of failure due to creep cavity area at various strains is calculated using Equation (18). It is well approximated with the aid of Weibull distribution as shown in Figure 9(c,9d) for various true strain conditions. As expected, it is observed that the distributions shift towards the higher range of creep cavity with increase in strain.

The Kachanov's damage model is deterministic in nature, while prediction of failure is inherently a probabilistic problem due to associated scatter. Matic and Sruc [31] emphasized the need for probabilistic approach to describe the physics-based models and their corresponding parameters. The proposed probabilistic model could explain the creep characteristics and damage evolution of service exposed HP-40 grade of steel used in hydrogen reformer. It would be highly beneficial if estimation of remaining life of component could effectively be positioned within the scatter band of creep strain vs time plots. Nucleation of cavities being deterministic, the scatter band in damage evaluation under identical test conditions could be easily

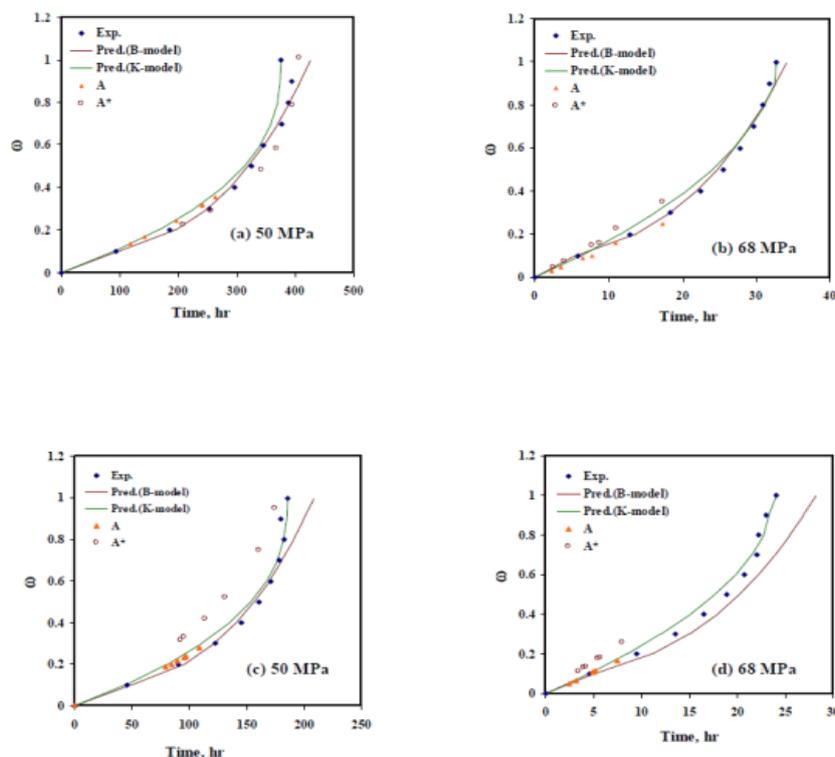


Figure 9b: Comparison of damage parameter A and A* with the damage prediction from probabilistic model as well as K-model at 870°C: (a,b) top and (c,d) bottom portion of service exposed reformer tube.

estimated by carrying out a Monte Carlo Simulation. This explains as to why such a large interval in scatter band was possible from both the models. It actually depicts estimation of all average times required to reach a specific damage state from both the models i.e. K- model as well as B- model [32,33].

The microstructurally quantified damage (creep cavitation /voids) is in close agreement with the damage prediction curves based on K-model as well as B-model, at low stress (50 MPa) whereas at high stress (68 MPa) it could correlate well till the mid region of the predicted curves, above which cavity measurement was practically not feasible (Figure 9ii). This is because at high temperature and at high stresses grain boundary sliding occurs discontinuously with time and amount of shear displacement is not uniform in the adjacent grains. But grain boundary at high temperature and high stresses can be accommodated by formation of folds at the grain boundaries [34]. When distribution of shear stress is non-uniform amongst the adjacent grains, at high stresses these folds may not be apparent as clear voids and possibly due to this fact void quantification under light optical microscope and SEM was only possible till the mid portion of the predicted curves (Figure 9ii). It is observed that the A* is more sensitive to life prediction than A because it picks up the localized damage in the form of voids which actually forms in the tertiary region of creep life.

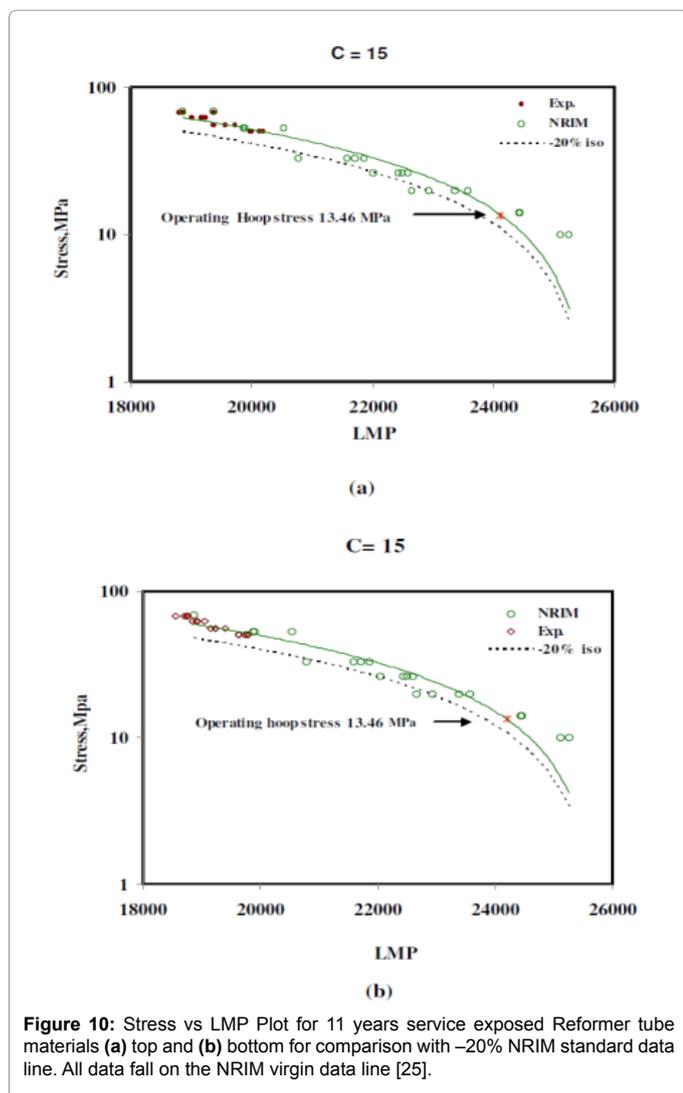
Residual life prediction of service exposed reformer tubes

There was no evidence of creep cavitation and voids in the service exposed tubes in as received condition, therefore only as far as creep strength and residual life at 870°C /13.46 MPa is concerned, there is no appreciable degradation due to service exposure and it is justified that it possible to obtain an additional minimum balance life of > 10 years

for both the top as well as bottom portions of the reformer tubes in the present investigation. However, at the design temperature of 920°C /13.46 MPa an additional minimum balance life of 7 years and 6 years for the top and bottom portions of the 11 years' service exposed primary reformer tubes respectively is possible from Stress rupture plots (Figure 10 and Table1), provided there is no evidence of any localised damage in the form of surface cracks, cavitation or dents and overheating of the tubes resulting in early onset of creep damage. Another check for safety of the service exposed tubes, similar to the current work, in terms of residual life is recommended to be carried out after expiry of about 2.5 years of service life from the view of economical and safety reasons. Also during shut down of the plant, NDT (non-destructive) tests viz. Dimensional (thickness and diameter) measurement, hardness measurement and in situ metallography may be carried out to assess the condition of the materials for their future serviceability.

Life assessment of 55 years' service exposed CCU reactor and FPU distillation column materials from accelerated creep/ stress rupture tests, creep damage assessment and LMP equation

Optical microscopy was carried out on as received as well as creep fractured specimens of 55 years' service exposed reactor and column material. Details of the procedure are reported and discussed elsewhere [15]. For as received material, samples were observed thickness wise, whereas, for crept specimens, samples were cut parallel to stress axis starting from fracture tip and mechanically polished with various grades of emery paper. The microstructures are shown in Figure 11,12. Detailed microstructural studies of as received material and fractured creep specimens as reported in [15] have revealed following features:

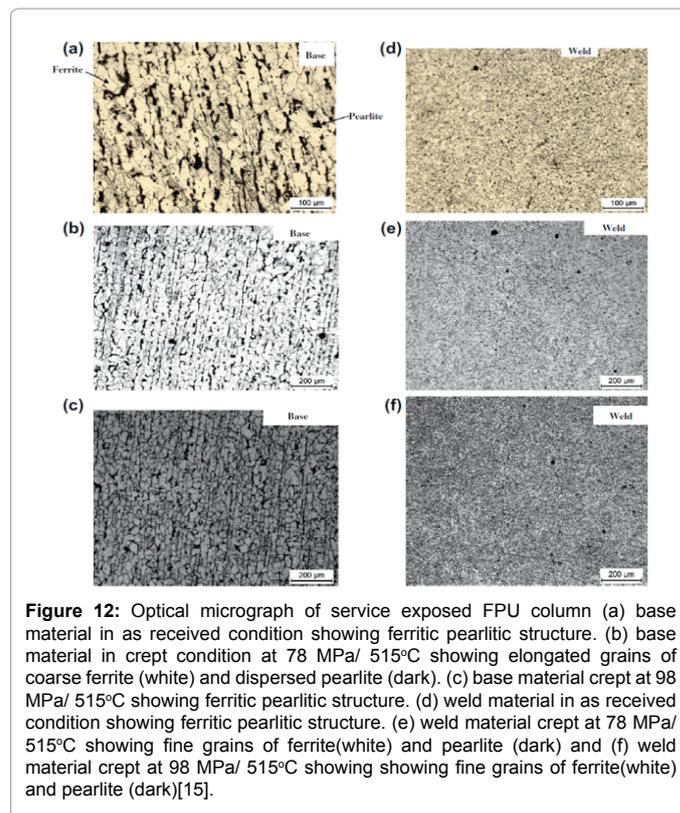
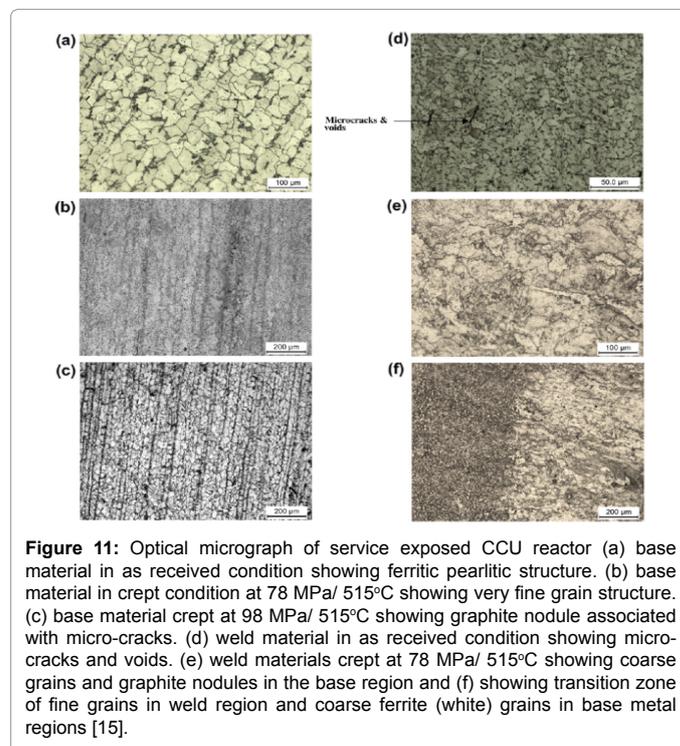


(i) It was noteworthy that creep cavities varied between 0.25-0.5% and 0.096-0.21% in crept specimens of service exposed CCU reactor base and weld material respectively. However, no creep cavity was observed in as-received CCU reactor base and weld material. Variation in percentage graphitization was between 6-9% and almost nil in crept specimens of service exposed CCU reactor base and weld material respectively. Also, 6% and almost nil graphite were observed in as-received CCU reactor base and weld material respectively. There was marginal increase both in creep cavitation and percentage graphitization for service exposed crept CCU reactor specimens compared to as-received service exposed CCU reactor specimens.

(ii) No creep cavity was observed both in as-received and crept specimens of service exposed FPU column base material. But, as-received FPU weld material had only around 0.06% voids. However, crept FPU weld material had 0.095-0.25% creep voids.

(iii) Variation in percentage graphitization was between 9-16% and 3-5% in crept specimens [15] of service exposed FPU column base and weld material respectively. However, ~8% and ~3% graphite were observed in as-received FPU column base and weld material respectively. A similar trend in marginal increase both in creep cavitation and percentage graphitization as of crept CCU reactor

specimens was observed for service exposed crept FPU column specimens too, compared to as-received service exposed CCU reactor and FPU column specimens respectively. Quantification of creep cavitation as well as graphitization was made with the help of Olympus Image Analyzer (Model: GX51) [15].



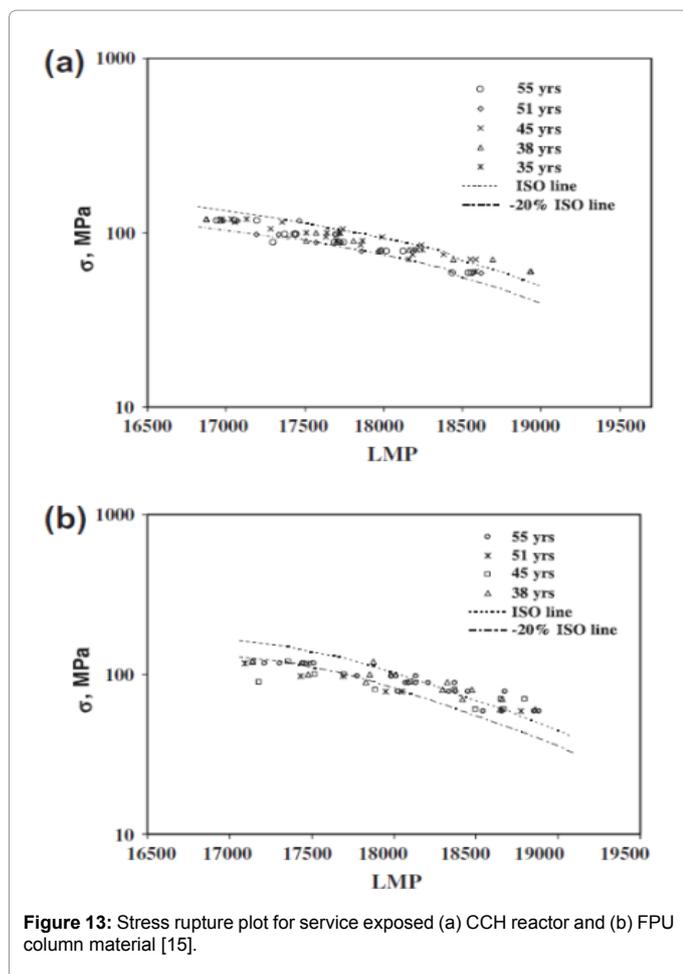


Figure 13: Stress rupture plot for service exposed (a) CCH reactor and (b) FPU column material [15].

Results of accelerated stress rupture tests conducted on reactor and column materials are revealed in Figure 13. The remaining creep life prediction of reactor as well as column materials was made using applied stress versus LMP curves. The plots contain the lower bound ISO curve for the respective grade of material for comparison. Because of uncertainty factors in materials and experimental conditions, creep rupture data generally exhibits dispersity, as shown in Figure 13a,13b, where the experimental data distributes scattering around the master curve in stress time–temperature plot. Uncertainty involved in calculating LMP (Larson Miller Parameter) value has been given prime importance because it is a function of rupture time. The placing of confidence bounds around any failure time prediction requires knowledge of the failure time distribution. In the case of the reactor, all weldment showed significant elongation indicating soundness of the weld [15]. However, for the column material, (Table 2), the weld metal showed poor creep ductility (the lowest being 4 %). A comparison of rupture data generated [15] with those obtained 4 years before, indicated that the rupture time of weld metal at any given stress had decreased 69% compared to the previous study (Table 2). At lower stress levels, both reactor and column materials showed a convergence with the standard ISO (master curve) curve. ISO curve, lower bound data is generally taken as the reference line for safe life prediction. Such an exercise had been performed in earlier investigations [33,34] as well for the same grade of materials. All base metal data fall on the lower bound ISO curve (–20% data curve) and weld metal data have marginally fallen below the ISO curve. This is usually anticipated that with increase

in service exposure, more creep damage is expected to accumulate in the material resulting in a decreased rupture time. However, the drastic reduction in creep ductility of the column weld is a matter of concern from the structural integrity point of view.

A comparison of experimental and predicted creep damage ω , using Equation (5) i.e.,

Material	Specimen	Temperature °C	Stress (Kg/mm ²)	Rupture time (h)	EL(%)	RA (%)
Ccu reactor	Base	515	12	44	28	53
	Base	515	12	32	40	49
	Base	515	10	111	31	47
	Base	515	10	138	36	41
	Base	515	9	307	29	36
	Base	515	9	282	29	44
	Base	515	8	738	23	35
	Base	515	8	1001	25	49
	Base	515	6	3529	15	33
	Base	515	6	2495	9	25
	Weld	515	12	66	7	16
	Weld	515	10	291	8	20
FPU-D1 column	Base	515	12	51	20	42
	Base	515	12	66	24	46
	Base	515	10	483	9	29
	Base	515	10	262	15	55
	Base	515	9	740	12	25
	Base	515	9	918	9.3	24
	Base	515	8	1327	5	20
	Base	515	8	3572	11	25
	Base	515	6	5971	8	34
	Base	515	6	2415	8	39
	Weld	515	12	99	10	31
	Weld	515	10	741	2.5	18
	Weld	515	9	1480	4	13
	Weld	515	8	537	2.1	12

Material	Specimen	Temp.(°C)	Stress, (Kg/mm ²)	Rupture Time, (h)	%EL (%)	RA (%)
CCU reactor	Weld	515	12	105	10	16
	Weld	515	10	204	6.3	22
	Cross weld	515	12	26	26	67
	Cross weld	515	10	48	18	43
	Cross weld	515	9	228	19	54
	Cross weld	515	8	333	16	46
	Cross weld	515	6	2608	9	28
	Base	515	6	3000	27	14
	Base	515	8	456	28	42
	Base	515	9	144	30	48
FPU-D1 Column	Base	515	10	72	31	47
	Base	515	12	34	29	48
	Base	515	12	36	43	78
	Base	515	10	96	33	76
	Base	515	8	576	38	72
	Base	515	6	4704	33	49
	Weld	515	12	110	22	70
	Weld	515	10	204	25	61
	Weld	515	8	432	31	38
Weld	515	6	4104	14	49	

Table 2: (a) Stress rupture properties of (a) 55 year and (b) 51 year service exposed CCU reactor and FPU column materials.

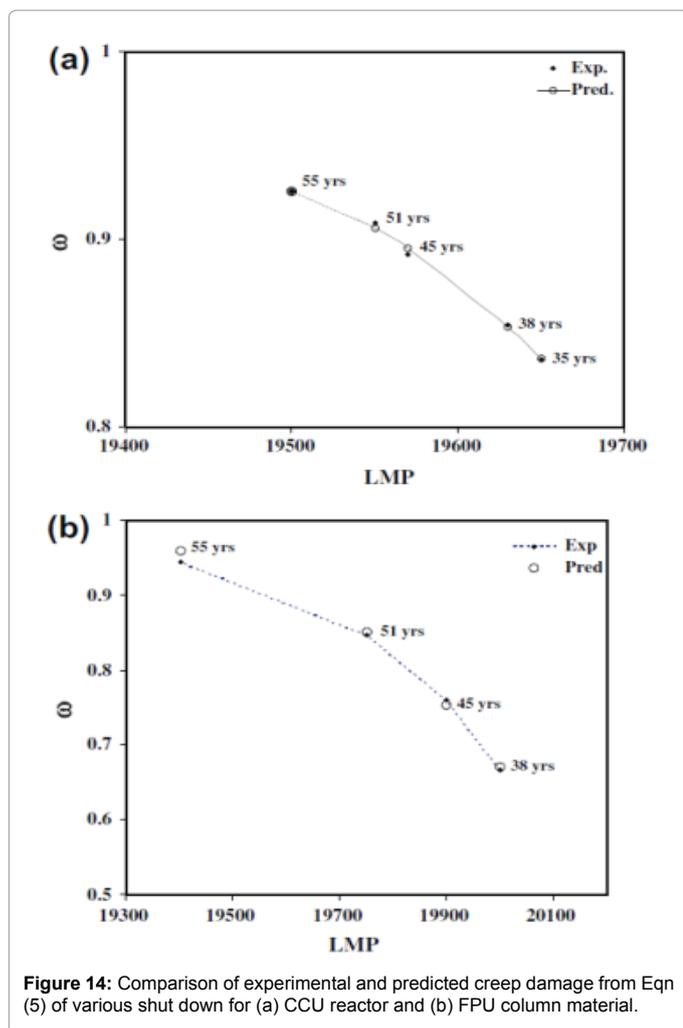


Figure 14: Comparison of experimental and predicted creep damage from Eqn (5) of various shut down for (a) CCU reactor and (b) FPU column material.

$$\omega = a_0 + a_1LMP + a_2LMP^2$$

,for various shut downs of (a) CCU reactor and (b) FPU column material has been made and is revealed in Figure 14. It depicts that with increase in service exposure time in years, ω was found to be on the higher side which is expected.

Uncertainty is involved in calculating LMP value because it is a function of rupture time. Figure 15 shows the effect of variation in LMP on Remaining Life (RL) calculation for reactor and column materials respectively, from the method reported in [15]. It has been well documented and reported in literature [7,23] that stress rupture tests always reveals scatter in rupture data. To incorporate data scatter band in stress rupture tests, Monte Carlo simulation has been carried out where LMP value is varied between ISO mean line and ISO (-20%). Corresponding to each LMP value, remaining life and normalized creep damage was calculated using Equations. (2)-(4). Variability of normalized creep damage using Equation (6) for reactor and column materials is well approximated with the aid of Weibull distribution as shown in Figure 15a,15b. It is observed that the distributions shift towards the higher range of damage with increase in service exposure time (Figure 16a,16b). Similarly, the probability of distribution of creep damage reveals that the distributions also shift towards the higher range of damage with increase in service exposure time for CCU reactor as well as FPU column material (Figure 17).

Remaining life of reactor and column materials after 55 years of service exposure with relative Confidence Interval (CI) are reported in Table 3. Monte-Carlo simulation could predict remaining life for exposed CCU reactor and FPU column materials with 99% confidence level. The lower and upper confidence interval indicates that 99% cases the remaining life will lie in this range with 99% certainty [15]. The reactor material needs replacement (Table 3) after a maximum service period of one year, as they are not in healthy state of health, in order to avoid plant accident resulting in huge loss of life and material property. The distillation column material can remain in service for at most three years and they should also be replaced for economic and safety reasons.

Future research would primarily be aimed at emphasizing on how best the design criteria could be met by real service conditions, possible ways to increase remnant life of similar components, improvements to the testing devices etc.

Conclusions

- i. Scatter observed in creep deformation and creep strain rate curve at 870°C of the top as well as bottom portion of the service exposed reformer tube at various stress conditions was probably due to variation in mode of fracture and scatter in voids.
- ii. From statistical point of view, the pattern of Weibull distribution followed for analyzing probability of rupture due to cavity or void area in the case of reformer tubes shifts with increase in

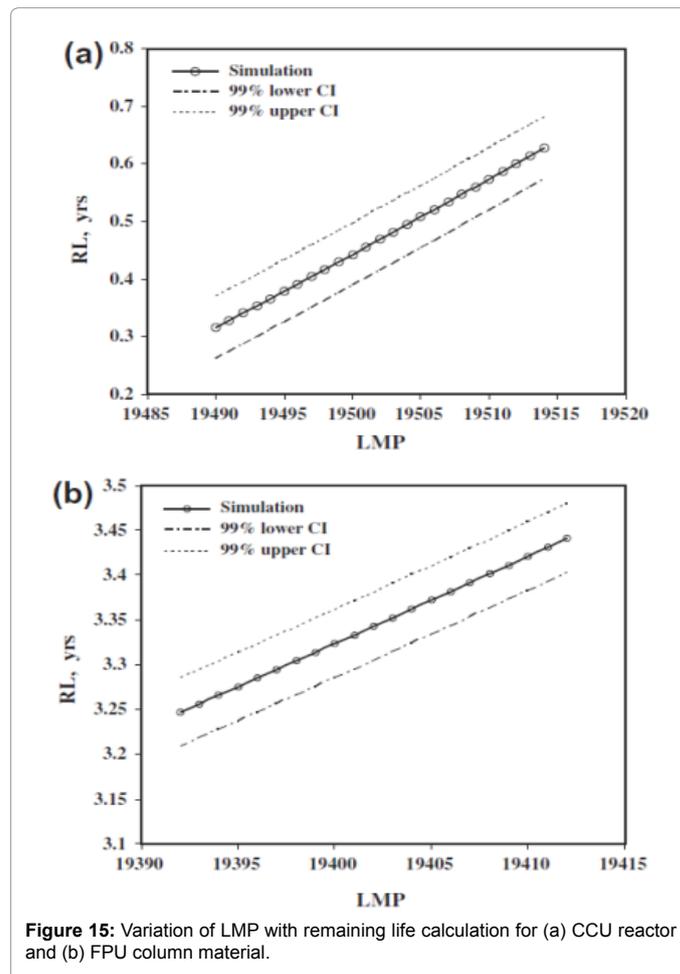


Figure 15: Variation of LMP with remaining life calculation for (a) CCU reactor and (b) FPU column material.

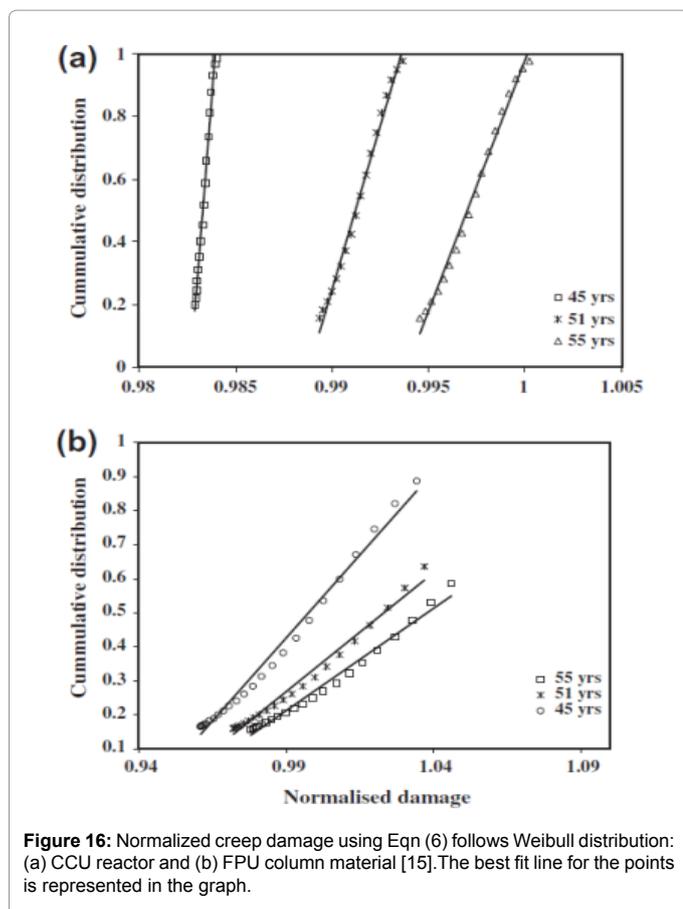


Figure 16: Normalized creep damage using Eqn (6) follows Weibull distribution: (a) CCU reactor and (b) FPU column material [15]. The best fit line for the points is represented in the graph.

true strain towards the higher population of void. For CCU reactor and FPU column materials, normalized creep damage could also be well-approximated with Weibull distribution which shifts towards the higher range of damage with increase in service exposure time.

- iii. Experimental data for estimating damage in the material, agrees well with simulation based CDM (Continuum Damage Mechanics) model as well as B-model for prediction of damage with scatter associated with it. In other words, estimation of average time to reach a specific damage state from both the models is in close agreement with that of experimental data. This pertains to the sudden changes in creep damage in the tertiary region as well.
- iv. Unlike at low stress, the microstructurally determined damage parameters A and A^* when compared with the damage prediction by the probabilistic model for the reformer tubes, clearly indicate that the prediction could only correlate till the mid region of the curves at high stress where cavity measurement was practically feasible. A^* is more sensitive to life prediction than A because it picks up the localized damage in the form of voids which actually forms in the tertiary region of creep life.
- v. A residual life of > 10 years is estimated at the operating stress-temperature conditions of the top as well as bottom portion of the service exposed primary reformer tube. This is realistic,

only if there is absence of any form of localised damage in the material. Monte-Carlo simulation could predict remaining life for exposed CCU reactor and FPU column materials with 99% confidence level. Based on accelerated creep rupture tests, service exposed CCU reactor and FPU column materials have balance lives of 2.4 and 4.78 years respectively. The distillation column material can remain in service for at most three years and they should also be replaced for economic and safety reasons. However, the reactor material needs replacement after a maximum service period of one year, as they are not in healthy state of health, in order to avoid plant accident resulting in huge loss of life and material property.

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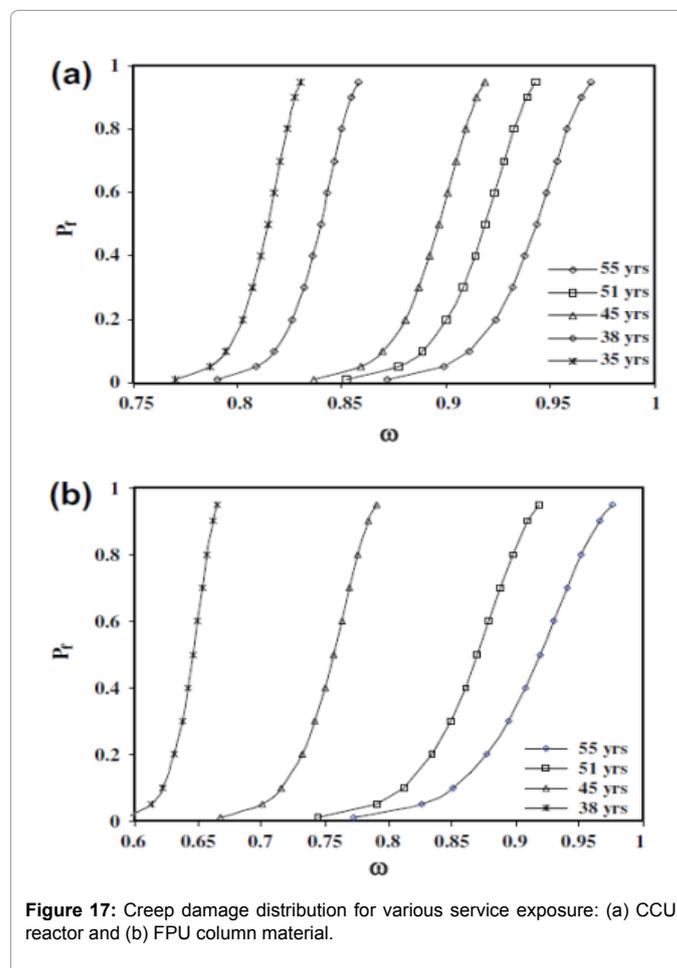


Figure 17: Creep damage distribution for various service exposure: (a) CCU reactor and (b) FPU column material.

Component (service exposed)	Temp. °C	RL, yrs	Confidence Interval (CI)	Lower CI	Upper CI
CCU reactor	515	2.4	99%	2.33	2.48
	520	0.47		0.42	0.52
FPU column	515	4.78	99%	4.73	4.83
	520	3.34		3.3	3.38

Table 3: Remaining life Assessment of 55 years service exposed CCU reactor (ASTM A201 Gr.A) and FPU column (ASTM A285 Gr.C) material with 99 % confidence limit.

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